DESIGN AND DEVELOPMENT

OF A HERMETICALLY SEALED

12 AMPERE-HOUR SILVER-CADMIUM CELL
IN A NON-MAGNETIC METALLIC CASE

FINAL REPORT

CONTRACT NO. NAS 5-2155

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

GODDARD SPACE FLIGHT CENTER
GREENBELT, MARYLAND



YARDNEY ELECTRIC CORPORATION

NEW YORK, NEW YORK 10013

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## **PURPOSE**

The purpose of this program was to design, develop and fabricate sealed silver-cadmium cells in non-magnetic metallic cases for tests under a satellite regime. The electrical power requirements called for a 12 ampere-hour cell to operate at 50% depth of discharge in a 100 minute cycle (charge - 65 minutes; discharge - 35 minutes).

## ABSTRACT

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This report describes the design, development and test results of hermetically sealed silver-cadmium cells built into non-magnetic metallic containers.

The final cell design was provided with an inner plastic case and an outer non-magnetic stainless steel housing. Special welding techniques for the manufacture of the stainless container were developed to avoid introducing magnetic properties.

The cells were grouped into five batteries of thirteen cells each and cycled by Boeing Company under a simulated 300-mile orbital regime at temperatures of -10°C; 20°C and 40°C, under vacuum conditions and using a radiation heat sink. Cell failures occured at 313 to 1350 cycles. (See Boeing report, Appendix (A)). Subsequently, epoxypotted cells of equal capacity made the same as the original YS-5-3(S) cells which gave up to 7000 cycles at 50% discharge were constructed and one battery of 12 cells and one battery of 3 cells were tested under the same orbital regime at 20°C. The 12-cell battery is still operating after over 4400 cycles. The 3-cell battery was discontinued due to a charger failure after 2371 cycles without cell failure.

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#### INTRODUCTION

This report concerns the design, development, fabrication and test evaluation of a hermetically sealed 12 ampere hour silver-cadmium cell in a non-magnetic metallic housing.

The contract specifically required that samples of the case materials be submitted to the Goddard Space Flight Center for evaluation and approval of the non-magnetic properties prior to the construction of the cells. It further required that prototype cells be filled with sufficient helium, to determine leakage rate, and be submitted to the same agency for approval prior to final production.

All design, development and engineering was carried out by Yardney Electric Corp., New York 13, New York. The battery evaluation testing, using a simulated satellite regime, was conducted by the Boeing Company, Seattle, Washington.

#### MECHANICAL DESIGN AND DEVELOPMENT

A material survey was conducted to find the metal best suited for the cell container housing. The survey specifications stated that among other properties the housing material must meet three main requirements, as follows:

It must possess low magnetic properties.

It must be able to be hermetically sealed to itself and other materials with relative ease.

It must be alkali-resistant.

As a result of the material survey certain types of stainless steels were selected as most suitable for the application. However, adequate hermetic sealing and manufacturing techniques were not available and required developmental time. Wetherefore decided to adopt an interim design consisting of a sealed plastic cell encased in a tin coated brass housing having a soft solder seal.

The mechanical design of the cell housing was approached by dividing it into two phases. Phase I (the interim design) was concerned with the design of the tin coated brass container around the plastic cell housing. Phase II (the final design) was concerned with the development, design and construction of the stainless steel container.

## Design Phase I

A total of 10 cell containers were fabricated from brass sheet, which were later tin coated to improve the soldering characteristics. The metal case covers were fabricated from tin coated brass sheet and were provided with two flanged openings to retain the ceramic ferrules for the cell terminals. All external current carrying hardware was fabricated from brass, which was silver plated and then gold flashed.

Two ceramic ferrules were used as "feed through" retainers for the extension terminals. The ceramic ferrules were metallized on their annular surfaces. The metallizing consisted of a thermally deposited molybdenum - manganese coating, with an electro-deposited layer of tin to improve the soldering characteristics.

## Cell Assembly Procedure

The plastic encased cell was potted into the metal case and the metal cover was soft soldered to the case. (The metal cover was preassembled by feeding the extension terminals through the ceramic ferrules and soft soldering them into position). The electrical contact was provided by means of set screws which were attached to the cell terminals and the extension terminals. To provide for air expansion during soldering, an opening of 1/8 inch was provided in the center of the cover. The opening was plugged and soldered after the completion

of all other soldering operations. The external surfaces of the metal container were teflon coated to provide a corrosion-resistant exterior finish.

#### Results of Design Phase I

The results of tests on design Phase I housing led us to the following conclusions:

- The soft soldering of the joints under tension does not provide a reliable hermetic seal.
- The attachment of the extension terminals by means of set screws is not reliable for good electrical contact.
- 3. The corrosion resistance of brass to alkali is poor and therefore the use of brass is undesirable even if the cell is first placed into a plastic container.

## Design Phase II

The information obtained from tests conducted on cells constructed under design Phase I, and from the techniques developed in the sealing of stainless steel materials, led to the construction of cells under design Phase II using non-magnetic stainless steel. To achieve greater reliability in hermetic sealing the stainless steel cases had to be welded to their covers.

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rejected by NASA as exceeding the maximum magnetic requirements, although in the annealed condition the material is non-magnetic. The search for a stable austenitic stainless steel, which would maintain its non-magnetic properties after working and welding, led us to type 310 stainless steel. A sample container was tested by NASA and found to possess less than the maximum permissible magnetic properties. Based on these results, stainless steel type 310 was chosen as the metal container for the final cells for this program.

The ceramic ferrules were modified in shape as shown in Dwg. #7945 Rev. B, to permit soldering to the cover and extension terminals after the cover had been welded to the case. The cell's plastic container inside the metal container, as adopted in design Phase I, was kept. The brass extension terminals were threaded to the cell terminals, eliminating the use of set screws.

The complete mechanical assembly features of this cell are shown in Dwg. No. 8007 Rev. C, and its component drawings.

The hermetic seal of each cell was tested by filling the cell with a gas mixture (85% oxygen, 15% helium) prior to sealing both the inner (plastic) and outer (metal) containers. Each cell was then tested for leakage using a mass spectrometer. All cells were found to possess a leakage rate of less than  $1 \times 10^{-6}$  cc per second, at a pressure differential of one atmosphere.

## ELECTROCHEMICAL DESIGN AND DEVELOPMENT

Three different cell designs were constructed and evaluated for this application. The differences consisted mainly in the plate surface area and quantities of active materials. These preliminary design evaluation models were encapsulated in potting but without metal housing for ease of construction. Prior to sealing, each cell was filled with a mixture of 85% oxygen and 15% helium.

The testing of these preliminary design evaluation models was done mainly to establish capacity, and capacity maintenance during cycling under a simulated satellite regime consisting of consecutive 65-minute charge and 35-minute discharge periods. At ten cycle intervals the cells were completely discharged to obtain the full cell capacity. The capacity results for the deep discharge cycles are shown in Table I. The results show that units of cell designs I and II still performed well after 125 cycles. Cell design III reached a low capacity within 65 cycles and testing was discontinued. Cell design II was chosen as the final design to be used in this program as a result of the superior electrical performance recorded in Table I.

100 cells were then fabricated based on the mechanical design

Phase II and the electrochemical cell design II, outlined previously.

This cell design has been designated as Yardney Electric Corp., sealed

Silcad® cell type YS12(M)S.

TABLE I

ELECTRICAL RESULTS OF THE PRELMINARY EVALUATION CELLS

TABULATION OF CAPACITIES FOR DEEP CYCLES ONLY \*

Cycle No.	Cell Design I	Cell D	esign II Cell #2	Cell Design III
5	12. 8	13.0	12.5	11.1
10	12. 3	13.2	12.4	10.5
15	12.7	-	-	9.8
20	11.6	-	-	-
30	10.6	13.7	13.1	-
40	-	13.3	12.0	8.7
50	10.0	14.1	13. 7	-
60	9.2	-	-	7.5 **
70	10.4	12.3	12. 5	
80	10.0	12.9	10.4	
90	8.8	13.4	13. 0	
100	8.9	10.5	12.5	
110	9.1	12.5	11.9	
125	9.2	12.5	12.1	
140		13, 3	13. 3	
150		13.8	13.8	

NOTES:

<sup>\*</sup> ALL CELLS WERE CYCLED AT THE SATELLITE REGIME OF 65-MINUTE CHARGE AND 35-MINUTE DISCHARGE. A FULL DISCHARGE CYCLE WAS APPLIED APPROXIMATELY EVERY 10 CYCLES. THE CURRENT USED FOR THE FULL CAPACITY DISCHARGE WAS 5 AMPERES TO 0.60 VOLTS PER CELL.

<sup>\*\*</sup> CELL DESIGN III WAS STOPPED AFTER 65 CYCLES BECAUSE IT FAILED TO MEET THE SATELLITE REGIME REQUIREMENTS.

#### BATTERY EVALUATION TESTS

These sealed 12 ampere-hour silver-cadmium cells were grouped into batteries and were tested, evaluated and reported on by the Boeing Company, Seattle, Washington. The complete tests description, results and analysis of results are reported in Boeing Document No. D 2-20496-1, "Final Report on Evaluation of Silver-Cadmium Batteries", (Appendix A).

The results can be summarized as follows:

The cells were assembled into five (5) groups of thirteen (13) cells each and cycled as a battery under simulated 300-mile orbital conditions. They discharged 35% of their nominal ampere-hour capacity per cycle and operated in radiation heat sinks at -10°C, +20°C and +40°C.

The first cell failures occurred at 313 cycles with others in the remaining batteries taking place at various points up to 1350 cycles. The early failures appeared to be related to leaking cells and to rapid loss in apparent capacity.

Subsequent to the early failures of these metal encased cells, two batteries of 12 AH epoxy-potted cells (12 and 3 cells respectively), manufactured and supplied by Yardney Electric Corp., were placed on test under the same cycling conditions. Those cells differed from the metal encased cells in that the separation material had been altered and the electrode surface area increased. They were scaled-up

duplicates of the YS-5-3(S) which gave good performance in previous Boeing tests. The twelve-cell battery is still operating normally in a 20°C radiation heat sink after over 4400 cycles. However, a charger malfunction caused failure of the three-cell battery on a 20°C conductive heat sink, after 2371 cycles without any cell failure.

#### CONCLUSION

Based upon the results obtained from the two types of cells, it is evident that the design chosen for the original 12 AH cell was not satisfactory for long cycle life. This was apparently due to the desire to obtain a minimum volume cell so that higher current densities and different separator systems were used than those proven out in the original YS-5-3(S) cells which were tested by Boeing, over the last three years. The 25% discharge group are still operating after 16,700 cycles.

It is apparent that this modification caused the generation of higher gas pressures during cycling which resulted in the subsequent rupture of the plastic cells. The rupture allowed leakage to the metal case and leakage between cells, causing an unbalance which led to premature cell failure.

Tests on capacity maintenance showed there was a greater capacity loss with the new design 12 AH cell than with the original design.

These are discussed in the Boeing report (Appendix A). It would appear that the combination of a different separator and higher current density caused a premature failure of these cells.

APPENDIX B

PARTS LIST FOR THE YARDNEY YS12(M)S SEALED CELL

Item II	Nomenclature	Dwg. or Part No.	Quantity per Battery
1	Metal Encased Cell Assembly YS12(M)S	YEC 8007C	1
2	Case	YEC 8013A	1
3	Cell Assembly	YEC 7952C	1
4	Cover	YEC 8014A	1
5 `	Ceramic Ferrule	YEC 7945B	2
6	Washer	YEC 7950B	2
7	Sockwasher	AN 935B-416L	2
8	Mex Nut	YEC 2710A-3	4
9	"Stimpson" Plug	*Stimpson No. D-3159	1
10	Spacer, Lucite . 032 x 3/16 x 1/4		8
11	Potting Compound		
	a) Epoxylite #211	Epoxylite Corp., El Monte, Calif.	35.4 g.
	b) Epoxylite Catalyst #1	11	2.3 g.
	c) Expoxylite Diluent #211-T	11	2.3 g.
12	Solder Washer, Rosin Cored 5/8 O.D. x 13/32 I.D. x .01	0 THK Commerical	<b>.</b> 2

Notes: \* Edwin B. Stimpson Company, 70 Franklin Avenue, Brooklyn, N. Y.

Yardney Electric Corp., New York

	REVISIONS		
SYM	DESCRIPTION OF THE PROPERTY OF	DATE	APPROVAL
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NOTE 9)  $\widehat{A_3}$ 

### NOTES:

- I. MAT LI STAINLEED STEEL TYPE BIO, AMMEALCD, . 025 THK, PER QQ-S-788.
- 2 FINITH: ALL EXTERIOR SUBFACES SHALL HAVE A FINITH EQUIVALENT TO A  $^{\pm}$ 3 POLISH OR BETTTER.
- 5 MFG. STDS PER YP-197.
- (A) 4. WELDING DIVELL CONFIDENT TO BRES. MILHARESS.

  WELDING BERNSCHESS SHALL SE INSET GAS

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  O'LIG (WHERE REQUIRED).
  - J. IDENTIFICATION MARKING PER YEC-928, CLASS "B RUBBER STAMP WITH PERMANENT INK.
  - FOR COVER SEE DWG NO. 8014
- CASE SHALL HOLD GAS PRESSURE AT 15 PSIG WITHOUT (A) LEAKAGE USE FIXTURE TOPREVENT CASE DISTORTION.
  - THIS PART IS USED FOR A NON-MASKETIC

    APPLICATION. NO MATERIALS OR PROCESSED

    SHALL BE SUPERITUTED OR ADDED WITHOUT

    PHIOR APPROVAL FROM YARDNEY ELECTRIC SOFT

    COLD WORKING SHALL BE KEPT TO A MINIMUM.
- 9. THE TWO DIMPLES SHALL STOP THE EDGE OF THE COVER A2. FLUSH WITH THE EDGE OF THE CASE, WHEN COVER IS ASSENTALED INTO THE CASE.

	REVISIONS							
ZONE	SYM	DESCRIPTION	DATE	APPROVAL				
	A	PREVD VIEWS AND CORR'D PICTURE 2) CHG'D ASS'Y PROCEDURES NOTES AND POTTING.  3) ITEM! WAS PARTNO BOIS CASE 518 ASS'Y 4) REVD ITEMS & E 10 5) ITEM 9 WAS SHIM. Q15 THK. ETHYL CELLULOSE 6) REMO ITEM 12- PATTING E 379 PER YP-327 7) TITLE OF DWG. WAS BATTERY HSSY FOR YS!2/M)S-1  ECN 698	10-30-62	De Co				
	В	1) ADDED NOTES G ? 7 2) REVD NOTE 4 3) ADDED CALL-OUTS OF NOTE 5 4) REVD CALL-OUT TOCEMENT OF ITEM 9 ECN-743	11-9-62	à 15				
20	C	DREVD NOTE & Z) NOTE 7 WAS G.5 GMS. ECN- 941	2-19-63	25/30				

E

# MOTES:

- 1. ASSEMBLY PROCEDURES:
  - Q. POUR IN THE REQUIRED AMOUNT OF POTTING (ITEM 10).

    PLACE CELL ASSEMBLY (ITEM 2) INTO CASE (ITEM I) ALLOWING

    POTTING COMPOUND TO DISTRIBUTE EQUALLY ON ALL SIDES.

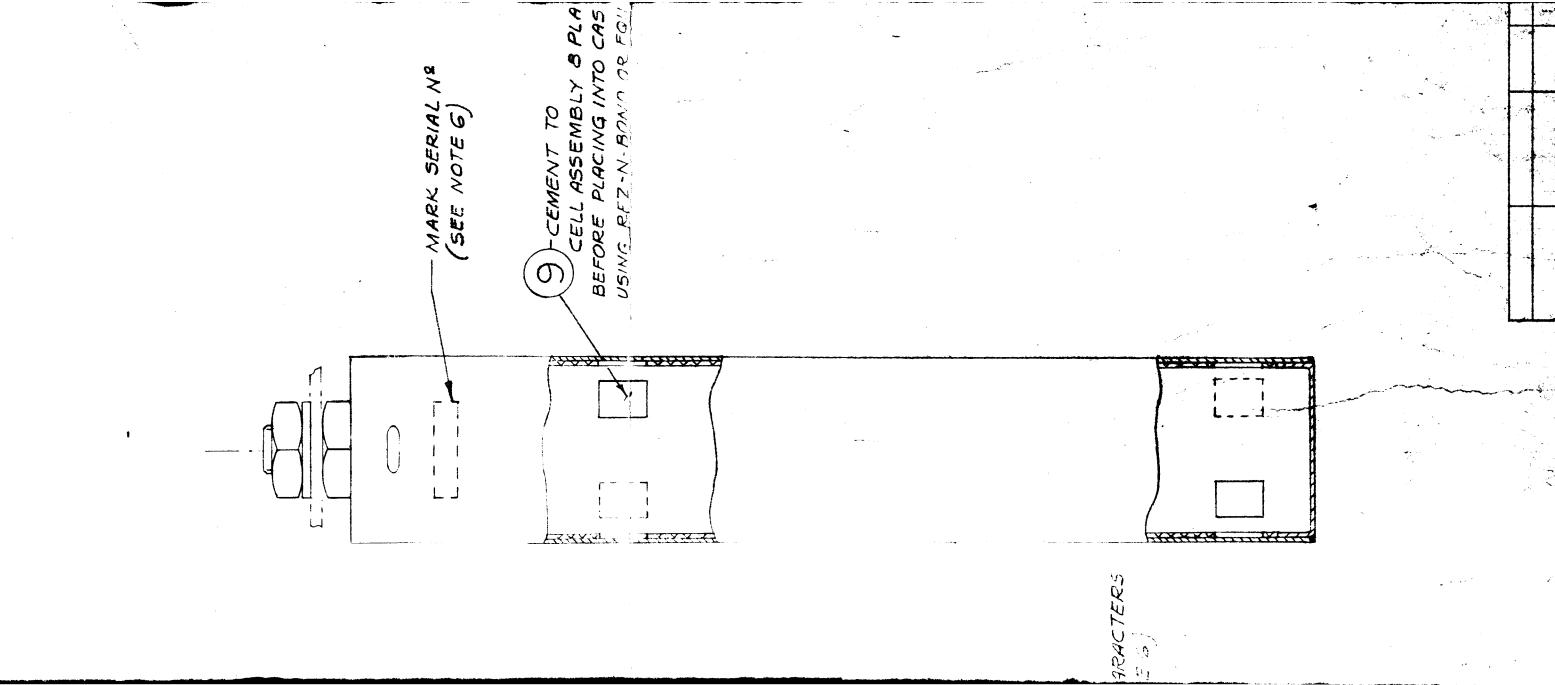
    CELL MUST BE PLACED FIRMLY AGAINST BOTTOM OF METAL CASE.
  - 6. USE APPROPRIATE FIXTURE TO CENTER EXTENTION TERMINALS WITH RESPECT TO CASE AND COVER AND ALLOW TO CURE.
  - C. USING EUTECTIC \*157 SOLDER, PRECORT OUTSIDE OF COVER TOP WITH A NEAT, EVEN CORT OF SOLDER. DO NOT ALLOW THE SOLDER TO RUN UP THE SIDES OF COVER.
  - d. PLACE COVER INTO CASE, FLUSH WITH TOP, AND WELD ALL AROUND (GAS TIGHT) PER MIL-W-8GII, TIG PROCESS . DO NOT ADD FILLER WIRE . IF FILLER WIRE IS REQUIRED TO SEAL CORNERS USE TYPE 310 ST. ST. WIRE ONLY. USE A SUITABLE HEAT SINK TO PROTECT THE SOLDER .
  - E. SOLDER, ITEM 4, TO COVER AND TERMINAL USING ITEM II
    AND ADDITIONAL GO LEAD 40 TIN, RESIN 44 COREGE, SOLDER
    WIRE, AS REQUIRED DO NOT ADD ADDITIONAL FLUXES.
  - f. SOLDER, ITEM 8, CLOSED USING SAME WIRE AS IN NOTE 1 e.
- 2. ASSEMBLE ITEMS 5,6 AND 7 AS SHOWN. 2. MANUFACTURING STANDARDS PER YP-197.
- 3. TORQUE BOTTOM NUTS 10-12 INCH. LBS.
- 4. ASSEMBLED CELL SHALL BE SUBJECTED TO A HELIUM LEAKAGE
  TEST AND SHALL HAVE A MAX LEAKAGE RATE OF IXIO'S CC/SEC. INA VACUUM.
- 5. THIS ASSEMBLY IS USED FOR A NON-MAGNETIC APPLICATION.

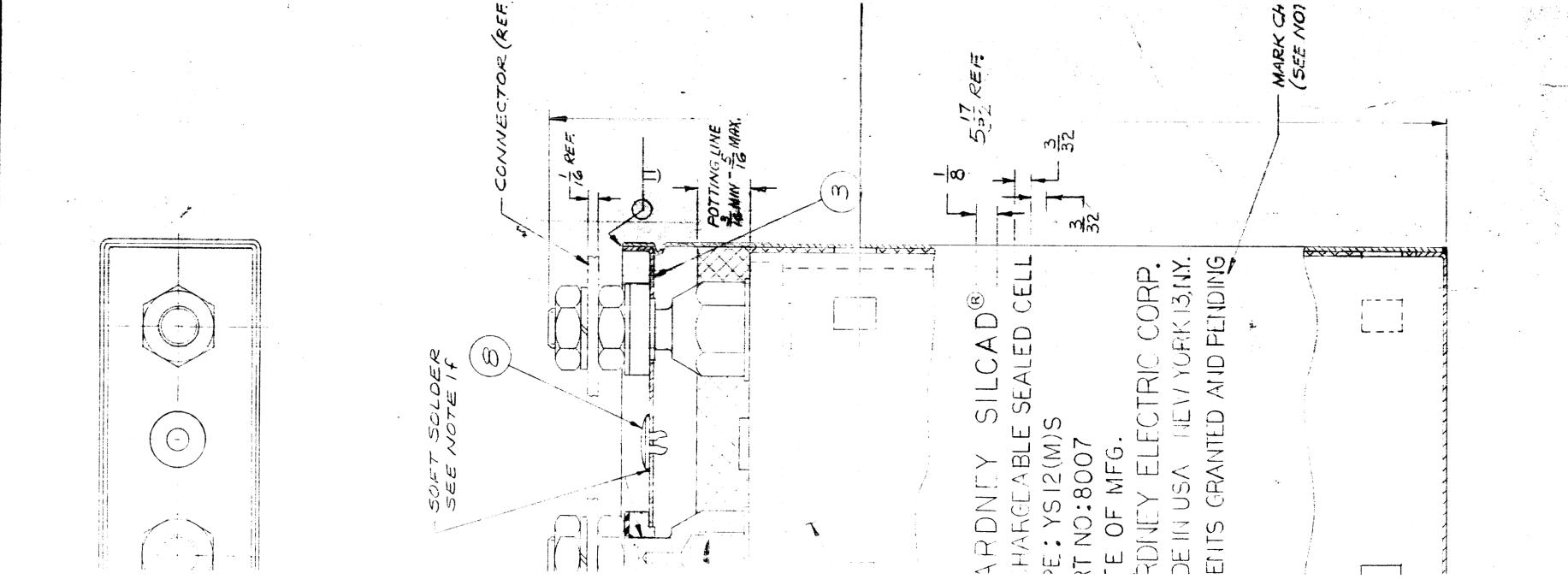
  NO MATERIALS OR PROCESSES SHALL BE SUBSTITUTED OR
  ADDED IN THE ASSEMBLY WITHOUT PRIOR APPROVAL FROM
  YARDNEY ELECTRIC CORP. ENG. DEPT.
- G. MARK CHARACTERS SHOWN WITH PERMANENT BLACK INK USING SILK SCREEN OR RUBBER STAMP, COVER AREAS WITH CLEAR LACQUER.
- 7. MIX POTTING COMPOUND I FEM 10, IN THE FOLLOWING

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10	40 GM5		POTTING, COMPOUND DILL			
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8	1	STIMPSO-1 AC. D 3157	"STHEPSON PLUS	BRASS	SILVER PLATED	. 7
7	4	2710-3	HEX NUT	BR.455		
6	2	AN 9358-4,6L	LOCKWASHER	BLON'ZE	SILVER PLATE FER SIG-S-365	- Agricial
5	2	7950	WASHER	NYLON		
4	2	7945	CENTINIC FERRULE			
3	1	8014	CO.IER	The second secon		
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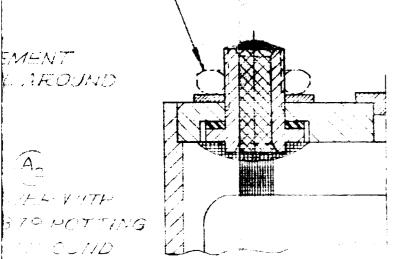




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SEE NOTE 5.
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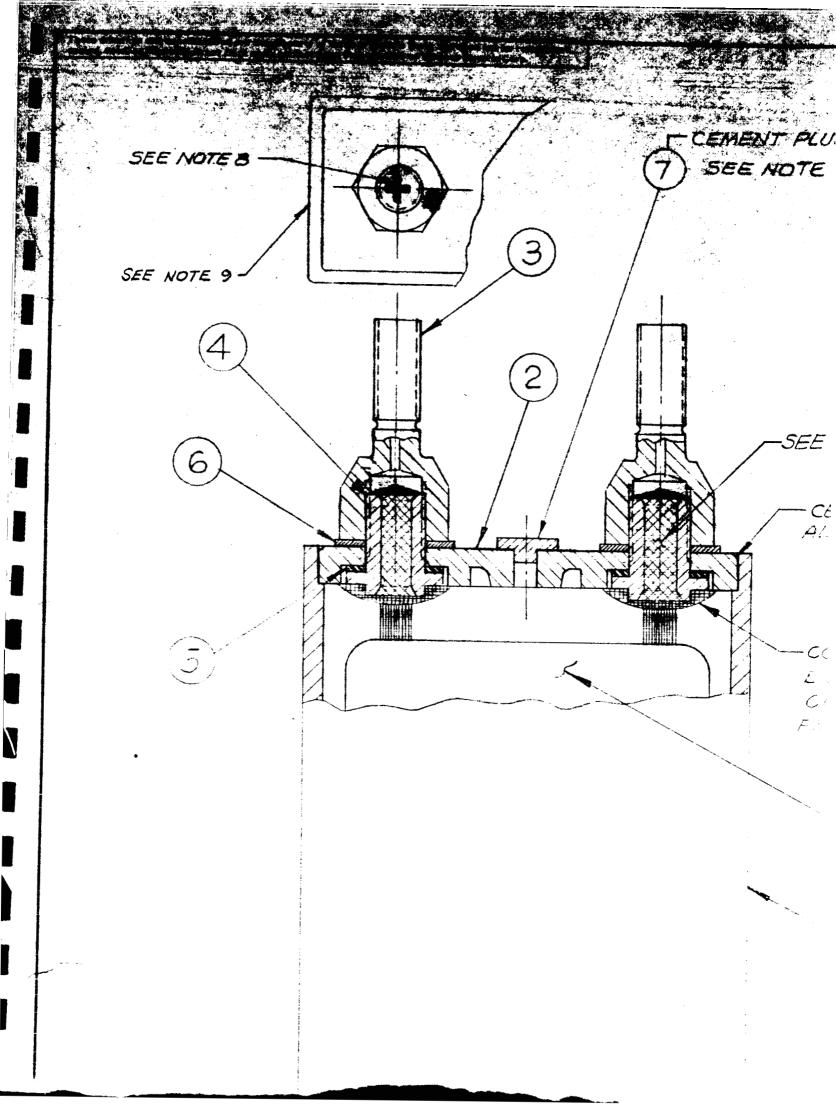
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- 2. ASSEMBLE AND CEMENT COVER AS SHOWN.
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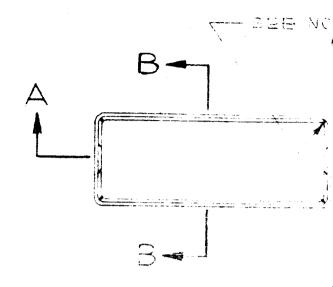
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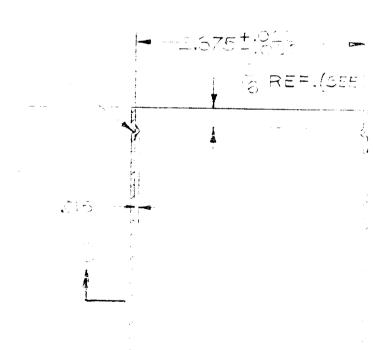
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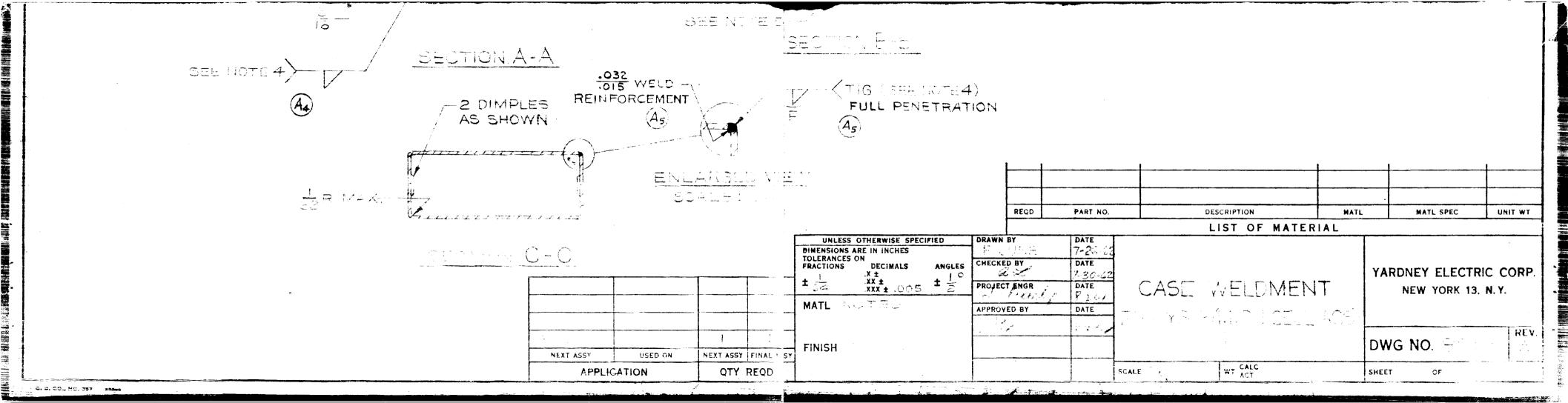


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#### NOTES:

I. MAT L: SST TYPE SIO, ANNEALED, 025 THK. PER QQ-S-783.

SEE NOTE 5)

- (A)2.
- FINICH: POLISH EXTERNAL SURFACES TO A"3 FINISH OR BETTER.
- 3. MF3. CTDS PER YP-197.

NOTE S)

- A.
- WELDING SHALL CONFORM TO SPEC. MILW-8611. WELDING PROCESS SHALL BE INERT-GAS TUNGSTEN ARC USING FILLER WIRE TYPE 310 (WHERE REQUIRED).
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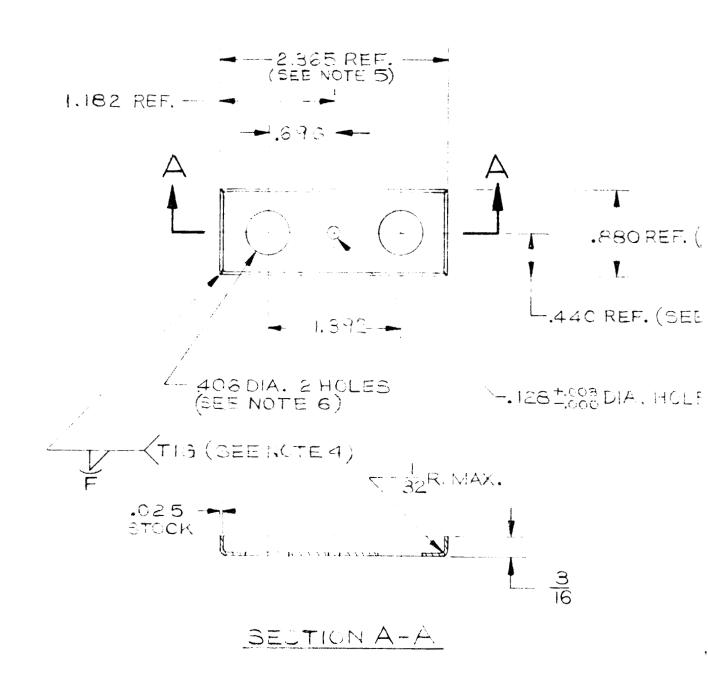
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- 7. THIS PART IS USED FOR NON-MAGNITIC APPLICATION.

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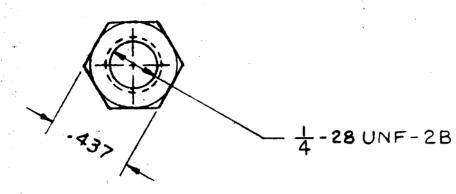
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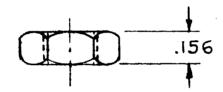
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- 2. HOLE SIZE AFTER METALLIZING.
- 3. MFG. STD'S PER YP-197
- 4. ALL MATERIALS SHALL BE NON-MAGNETIC.

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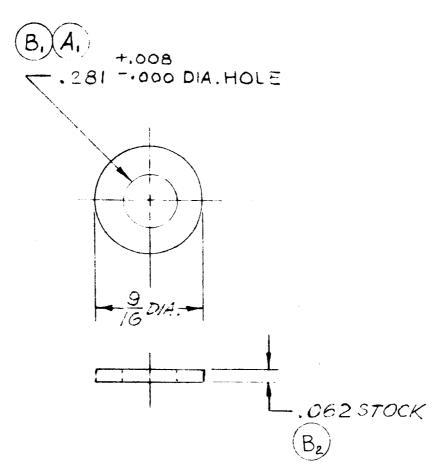
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PARTINO	N	MATERIAL		FINISH	
2710	C.R.5 PER QQ-5-633, F5-B 1112			SILVER PLATE PER QQ-5-365, TYPE III, (:0002 THICK.)	
2710-3	QQ-B-	55 ROD PER 626,COMP. ARD TEMP.	22	GOLD PLATE PER MIL-G-14548A TYPE II, CLASS I (.00005 THICK) OVER SILVER PLATING PER QQ-5-365, TYPE III 10002 THICK	
UNLESS OTHERWIS	E SPECIFIED	DRAWN BY	DATE	TITLE	
DIMENSIONS ARE IN TOLERANCES ON: FRACTIONS DECIMA .X.± ± 4 .XX ± .XXX±. MATL NOTED	±	A. KRASSILNIKOVA CHECKED BY  PROJ ENGR  APPROVED BY	DATE 3-/-6 DATE 4-4-6/	NUT, HEX \(\frac{1}{4} - 28\)	YARDNEY ELECTRIC COMP NEW YORK 13, 14 %
Trinian MOTE	<b>u</b>		354	SCALE 2/1	

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I	APPLICATION			REVISIONS					
	NEXT ASSY USED ON		SYN DESCRIPTION		DATE	APPROVAL			
Γ	8007		Α	WAS 1) 17/64 DIA 2) NEXTASSY 1956(2) 3) USED ON 1956(2) FOISGI	8-3-62	a to co			
			В	WAS 1) 312 + 1008 - 1000 DIA HOLE 2) 1032 STOCK ECN" 694	10-26-62	C C to			



NOTE:

MFG. STDS. PER YP-197.

UNLESS OTHERWISE SPECIFIED	DRAWN BY	DATE	TITLE			
DIMENSIONS ARE IN INCHES TOLERANCES ON: FRACTIONS DECIMALS ANGLES  .X±  .XX±  .XXX±  .	C.VASSIMEPP CHECKED BY PROJENGR APPROVED BY	5-25-62 DATE 6-5-62 DATE AND W			YARDNEY ELECTRIC CORP. NEW YORK 13, N.Y.	
FINISH		t. ' ' t	SCALE 2:/	DWG No.	7 <i>950</i>	rev. B